## **CSIR**

## Defence, Peace, Safety and Security

# Aircraft vulnerability analysis by modelling and simulation

Nelis Willers, Riana Willers & Alta de Waal Presented by: Kevin Gopaul

AOC Aardvark Roost Mini Conference

17 November 2014



#### "Science advances one funeral at a time"

Max Planck (1858 – 1947)

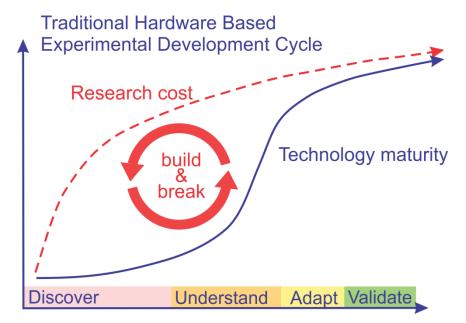


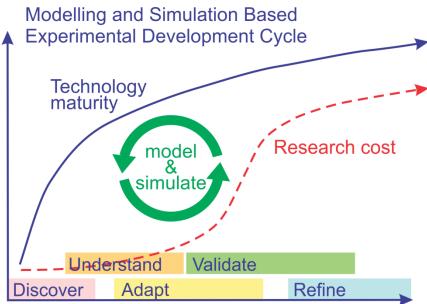
## **Background**

- Introducing various concepts
- The information in this presentation
  - you have seen before, and
  - is publicly available
- You will, however,
  - see it in a different way, and
  - hopefully acquire one or two new insights
- Security
  - Models used are desensitised (some available online)
  - Illustrate the thesis of this paper
- Significant modelling and computational effort only some of the results are reported here

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#### Introduction





- "Build and break" hardware.
- Labour, time & cost intensive.
- Workflow highly linear.

- Design by software changes.
- Experiment at lower cost & risk.
- Workflow in parallel paths.

**Graphics adapted from Airbus Industries pamphlet** 



## Introduction

- The software/simulation must be:
  - representative,
  - accurate, and
  - validated
- We do not want:



# Top 20 Replies by Programmers when their programs don't work...

- > 20. That's weird...
  - 19. It's never done that before.
- 18. It worked yesterday.
  - 17. How is that possible?
  - 16. It must be a hardware problem.
  - 15. What did you type in wrong to get it to crash?
  - 14. There has to be something funky in your data.
  - 13. I haven't touched that module in weeks!
  - 12. You must have the wrong version.
  - 11. It's just some unlucky coincidence.
  - 10. I can't test everything!
    - 9. THIS can't be the source of THAT.
    - 8. It works, but it hasn't been tested.
    - 7. Somebody must have changed my code.
    - 6. Did you check for a virus on your system?
    - 5. Even though it doesn't work, how does it feel?
    - 4. You can't use that version on your system.
    - 3. Why do you want to do it that way?
  - 2. Where were you when the program blew up?
  - 1. It works on my machine.



## **Overview**

- Optical signatures
- 3D background radiance
- Hemispherical irradiance
- Aircraft model and signature
- Missile model
- Miss distance prediction
- Miss distance data fusion

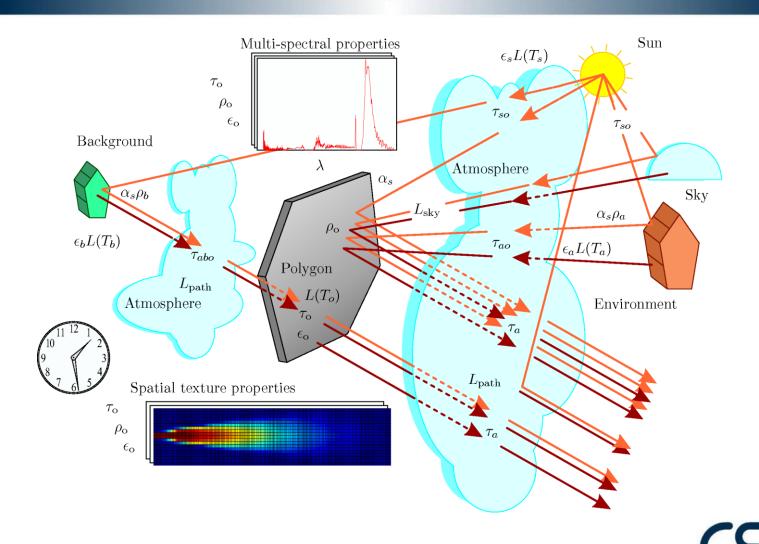


## **Optical Signatures**

- An infrared signature depends on many factors:
  - Shape and size of the object
  - Temperature and emissivity
  - Surface reflection of external sources (earth, sun, sky)
  - Transmittance through the object
  - Background against which (and atmosphere through which) it is viewed
- Apparent signature changes with changes in:
  - Weather and atmospheric conditions
  - Time of day
  - External and internal heat sources



## **Optical Signatures – Signature Rendering Model**



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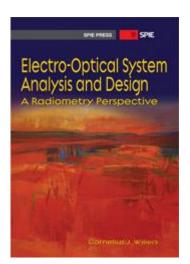
## **Optical Signatures**

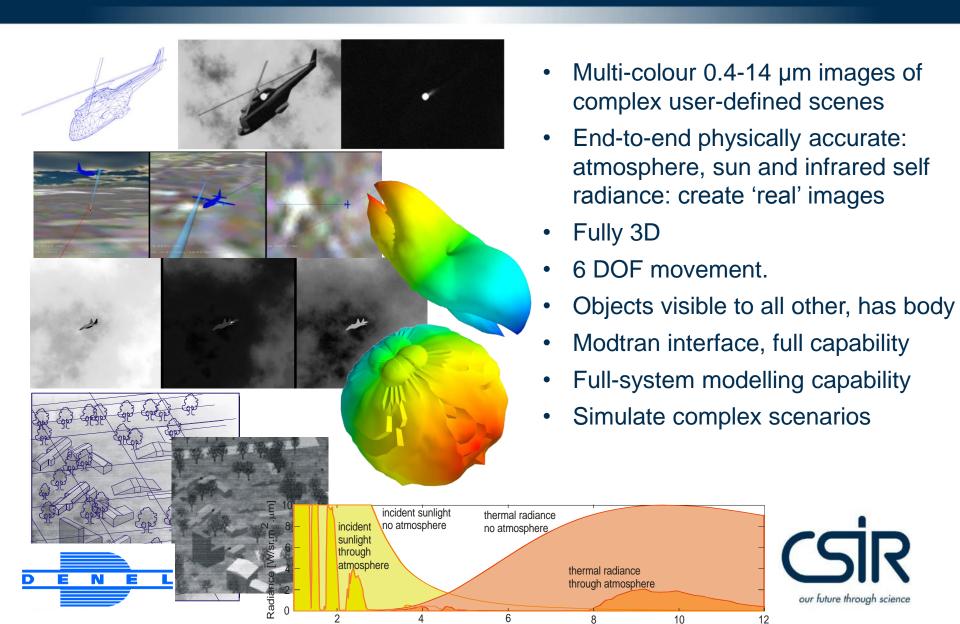
thermally emitted 
$$L_{\rm self}$$
 transmitted background  $L_{\rm trn\ back}$  
$$L_{\mathcal{S}} = \Delta_{\epsilon} \int_{0}^{\infty} \epsilon_{o\lambda}(\theta_{v}) L_{\lambda}(T_{o}) \tau_{a\lambda} \mathcal{S}_{\lambda} d\lambda + \int_{0}^{\infty} \tau_{o\lambda} \epsilon_{b\lambda} L_{\lambda}(T_{b}) \tau_{abo\lambda} \tau_{a\lambda} \mathcal{S}_{\lambda} d\lambda$$
diffuse reflected ambient background  $L_{\rm ref\ amb}$  diffuse reflected sky  $L_{\rm ref\ sky}$ 

$$+ \Delta_{\rho} \int_{0}^{\infty} \int_{\rm env} \rho_{o\lambda} \epsilon_{a\lambda} L_{\lambda}(T_{a}) \tau_{ao\lambda} \tau_{a\lambda} \mathcal{S}_{\lambda} d\Omega d\lambda + \Delta_{\rho} \cos \theta_{a} \int_{0}^{\infty} \int_{\rm sky} \rho_{o\lambda} L_{\rm sky} \lambda \tau_{a\lambda} \mathcal{S}_{\lambda} d\Omega d\lambda$$
reflected sun  $L_{\rm ref\ sun}$  atmospheric path radiance  $L_{\rm path}$ 

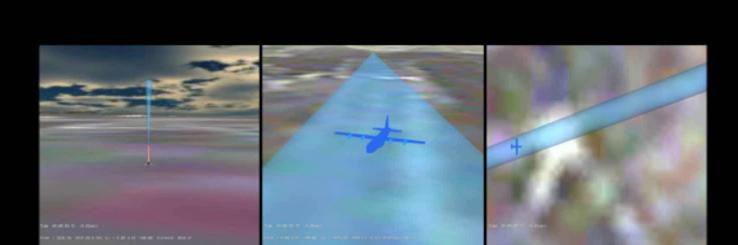
$$+ \Delta_{\rho} \psi \cos \theta_{s} \int_{0}^{\infty} f_{r}(\theta_{i}, \theta_{s}, \varphi_{i}, \varphi_{s}) \epsilon_{s\lambda} L_{\lambda}(T_{s}) \tau_{so\lambda} \tau_{a\lambda} \mathcal{S}_{\lambda} d\lambda + \int_{0}^{\infty} L_{\rm path} \lambda \mathcal{S}_{\lambda} d\lambda$$

(Willers 2013)





Aircraft Vulnerability Simulation



Aircraft Vulnerability Simulation



Aircraft Vulnerability Simulation

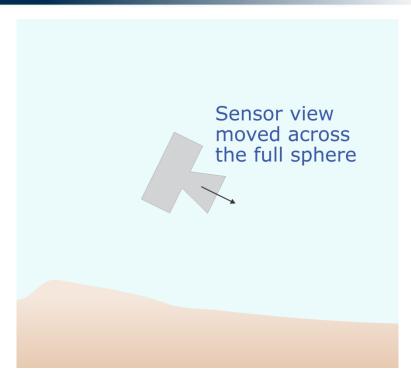


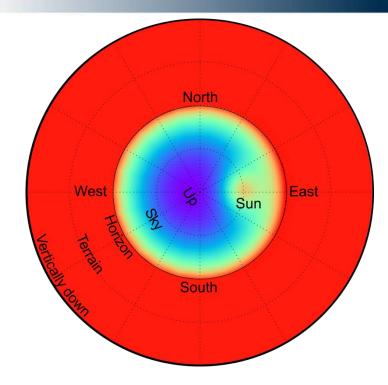
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## Background Radiance (Skydome)



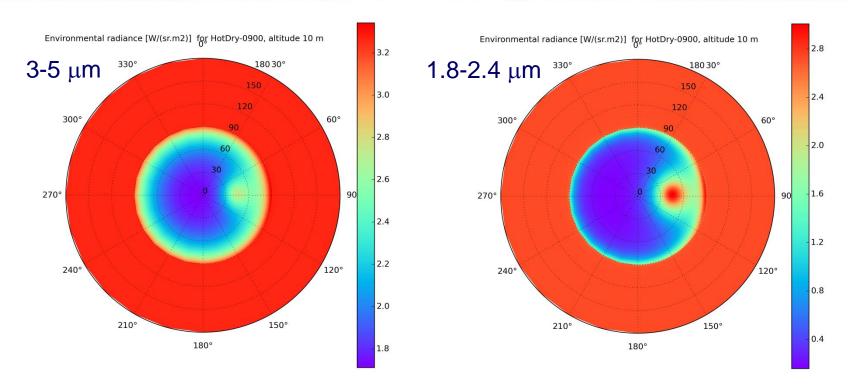


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- Use Modtran to calculate radiance for all views over spheres sky (up), horizon and terrain (down)
- Sensor image painted with skydome first and then paint the target and scene radiance in front of skydome
- Missile sees this background too

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## **Background Radiance (Skydome)**



09:00 in `Desert Extinction' aerosol model, 10 km visibility, 10 m alt

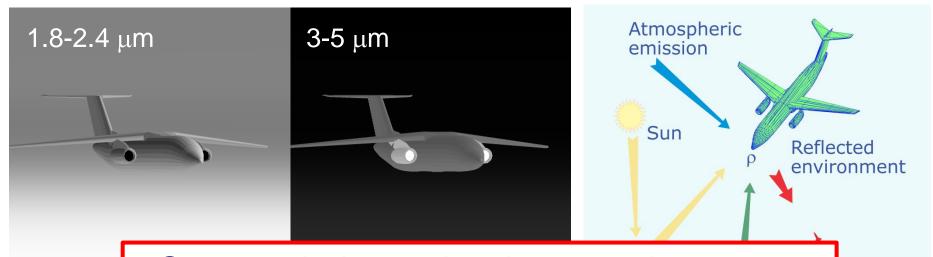
The target intensity is seen as a contrast against this radiance.

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## Hemispherical Irradiance – Earth Shine and Sky Shine



Over and above the thermal signature, the target also radiates this reflective flux.

- Earth shine and sky shine irradiate the target
- Significant effect if there is no visible hot metal or open plume
- Could provide positive contrast against dark sky in SWIR
- Especially important for near-zero contrast scenarios

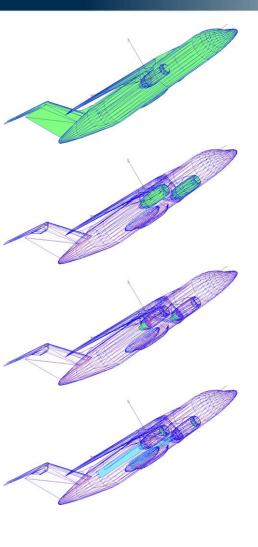


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## **Aircraft Model and Signature**



- Antonov 148
- 30 m wingspan
- 80 seats (15 tons)
- Twin turbofan
- Wireframe purchased online

#### Geometric model

- 2871 polygons
- 10 poly classes
- Plume

## Radiometric model

- Emissivity
- Reflectance
- Temperature



## Aircraft Model and Signature – Engine EGT

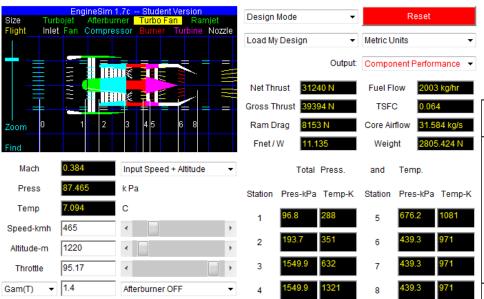
# NASA

#### EngineSim Version 1.7c

#### Glenn Research Center

http://www.grc.nasa.gov/WWW/k-12/airplane/ngnsim.html

This is a beta 1.7c version of the EngineSim program, and you are invited to participate in the beta testing. If you find errors in the program or would like to suggest improvements, please send an e-mail to Thomas.J.Benson@nasa.gov.



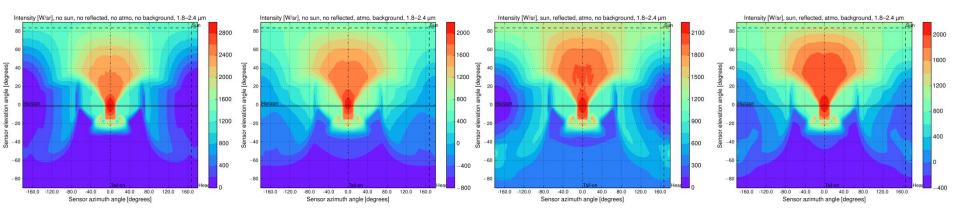


	Throttle	Altitude [m]						
	setting	0.0	1220.0	3000.0	5000.0	8000.0	12000.0	19800.0
	100.0	980.0	988.6	1001.9	1016.2	1038.0	1060.8	1060.8
	98.0	953.4	972.4	976.2	990.5	1012.4	1035.2	1035.2
	95.0	912.5	923.0	934.4	949.6	971.5	993.3	993.3
	82.0	731.9	741.4	754.7	770.0	792.8	815.6	815.6
Ш	73.0	616.9	616.9	630.2	645.4	669.2	692.0	692.0
	70.0	565.6	575.1	588.4	603.6	627.4	651.1	651.1
	68.0	537.1	549.4	560.8	576.0	599.8	622.6	622.6
IJ	65.0	494.3	499.0	518.1	534.2	557.0	580.8	580.8
	%	K	K	K	K	K	K	K

- Turbofan exhaust gas temperature
  - Ground reference measurements
  - NASA model for missing data
  - Cross validation on measure points



## Aircraft Model and Signature – 1.8 – 2.4 µm



- Absolute intensity: thermal only 0 – 2800 W/sr
- Contrast intensity: thermal only -800 – 2100 W/sr
- Absolute intensity: comprehensive 0 2100 W/sr
- Contrast intensity: comprehensive -400 – 2100 W/sr

- 3D signature complex function of view
- Engine and fuselage geometry has significant effect
- All radiometry terms required to be included
- All radiometry terms required to be accurate



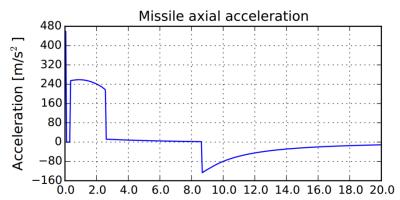
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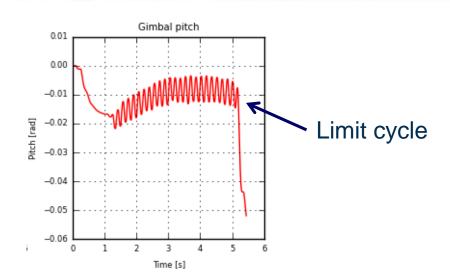


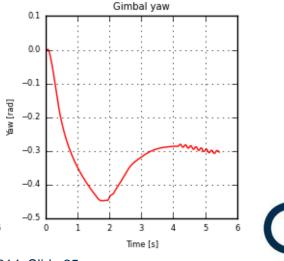
#### **Missile Model**

- Two-axis gimballed platform
- Maximum look angle of ±40 deg
- Tracking rate is limited to 20 deg/s
- Field of view 1.5 degrees
- NEE =  $2.0 \times 10^{-7} \text{ W/m}^2$
- $E_{threshold} = 1 \times 10^{-6} \text{ W/m}^2 \text{ (SNR=5)}$
- 1.8–2.4 µm spectral response
- Max airframe lateral acceleration 6g



Gimbal info: Birchenall, Richardson, Butters, and Walmsley, Infrared Physics & Technology 53, 372–380 (2010). Missile and sensitivity from first principles calculations









### Missile Model – Missile Centroid Tracker

#### **Contrast** irradiance (offset removal)

$$E(i,j)_{c} = ||E(i,j) - \langle E \rangle||$$

Select pixels exceeding a fixed threshold

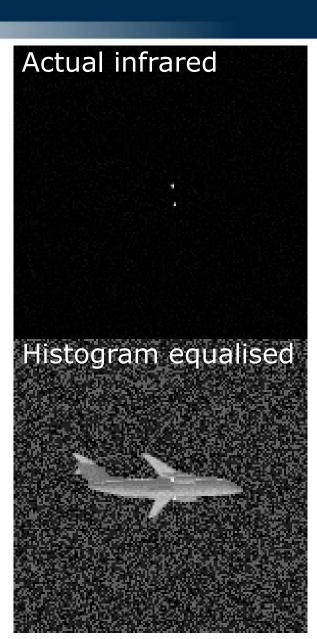
$$E(i,j)_{c} > E_{\theta}$$

#### **Centroid** of selected pixels

$$(\bar{i_t}, \bar{j_t}) = \left(\frac{\sum_{i} \sum_{j} iE(i, j)_c}{\sum_{i} \sum_{j} E(i, j)_c}, \frac{\sum_{i} \sum_{j} jE(i, j)_c}{\sum_{i} \sum_{j} E(i, j)_c}\right)$$

Target detected and tracked if

$$\sum_{i} \sum_{j} E(i,j)_{c} > 0$$



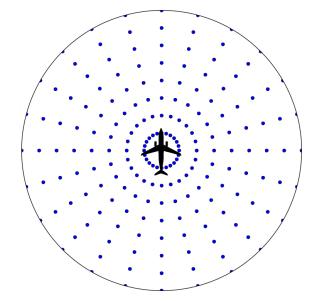
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#### **Miss Distance Prediction**

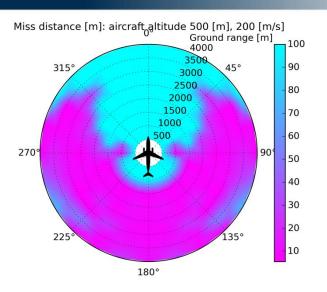
- In simulation, launch missiles at:
  - 15 degree intervals
  - 500 m intervals
- Find minimum distance between aircraft and missile

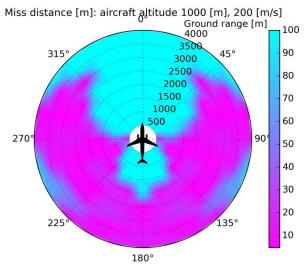


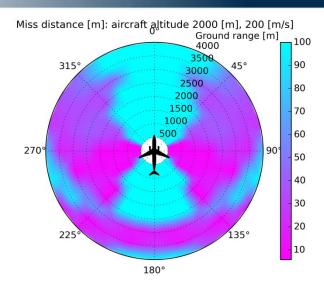
- Termination condition:
  - Miss distance increases
  - Target leaves sensor FOV
  - Target signal below threshold



#### **Miss Distance Prediction**







- Example
  - Constant altitude (500, 1000, 2000) m
  - Zero pitch
  - 90% throttle setting
- Colour → minimum distance between target and missile
- Countermeasure single MTV flare



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- Bayesian Network (BN)
  - Graphical models for reasoning under complexity and uncertainty
  - Marriage between probability theory and graph theory
  - Consists of nodes and edges
  - Nodes depict variables
  - Edges depict causal links between nodes
  - Different types of knowledge and data can be fused into one network
  - Handles missing data
  - Powerful inference engine

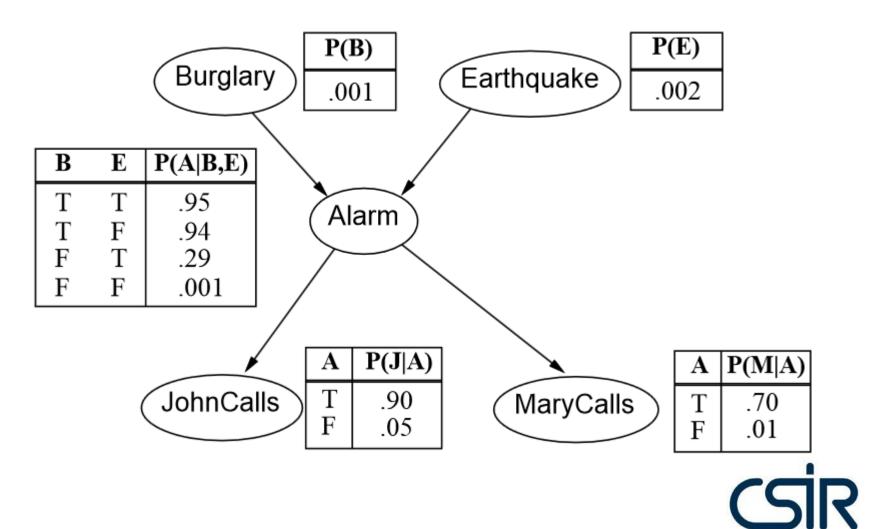


I'm at work, neighbor John calls to say my alarm is ringing, but neighbor Mary doesn't call. Sometimes it's set off by minor earthquakes. Is there a burglar?

Variables: Burglar, Earthquake, Alarm, JohnCalls, MaryCalls Network topology reflects "causal" knowledge:

- A burglar can set the alarm off
- An earthquake can set the alarm off
- The alarm can cause Mary to call
- The alarm can cause John to call

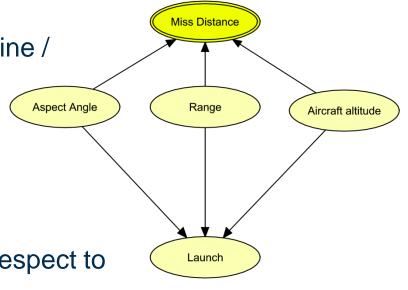




Aircraft vulnerability depends on

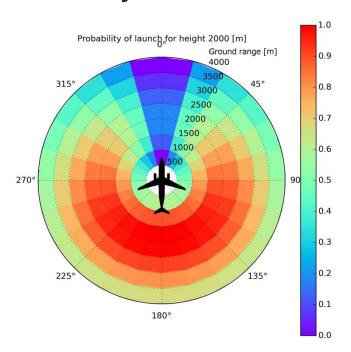
Probability of missile launch (doctrine / training), and

- Miss distance if launched
- Both have complex dependencies
- Simple model
  - Launch location aspect angle with respect to aircraft
  - Range between launch location and aircraft
  - Aircraft height above missile location
- A real-world production model has many more inputs



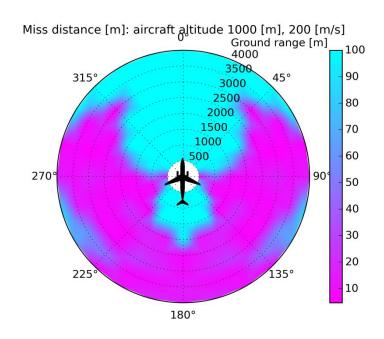


#### **Probability of missile launch**



- Training & doctrine
- Human expert panel
- This example by math model

#### Miss distance



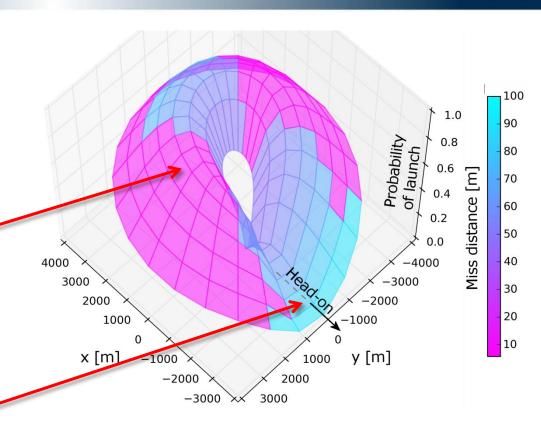
Simulation calculation



Simplistic and naïve approach combines the data:

- z-height: launch probability
- colour: miss distance

- High danger zones: magenta hills (small R<sub>miss</sub> and high P<sub>launch</sub>)
- Intermediate zones: cyan hills & magenta valleys.
- Safest zones:
   cyan valleys
   (large R<sub>miss</sub> and low P<sub>launch</sub>)

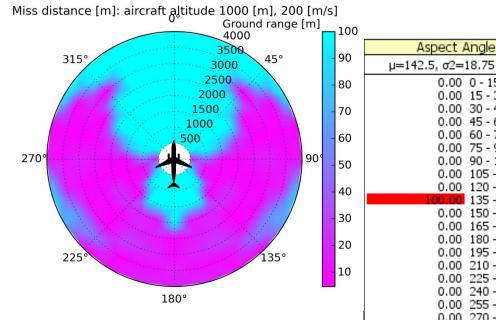


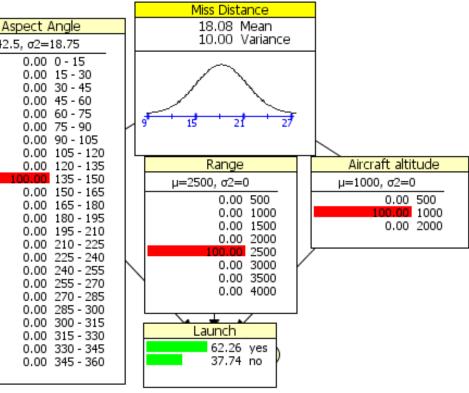


- Hugin BN modelling software
  - Developed specifically to develop BN for decision making
  - GUI
  - Inference Engine
    - Predictive Mode (top-down)
    - Prescriptive Mode (best state)
    - Diagnostic Mode (bottom-up (what if))
- Statistician
  - Dr Alta de Waal



#### Bayesian network provides **prediction** views on data



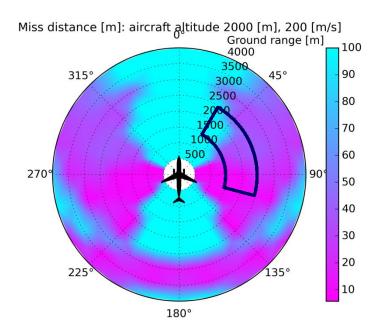


## This example shows

- miss distance and
- launch probability for single launch location

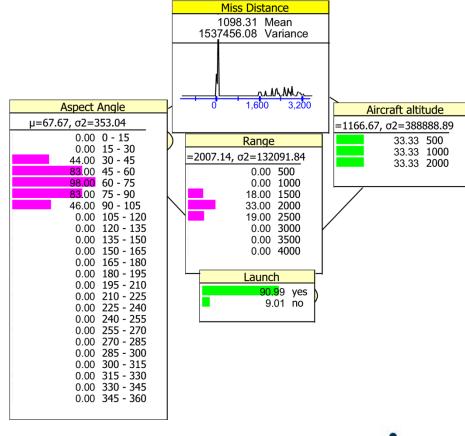


#### Bayesian network provides likelihood views on data



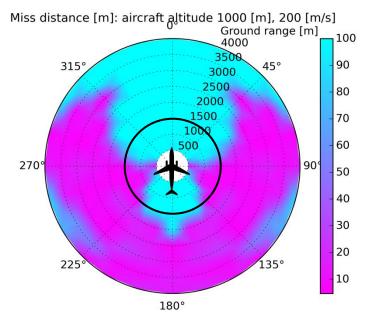
#### This example shows

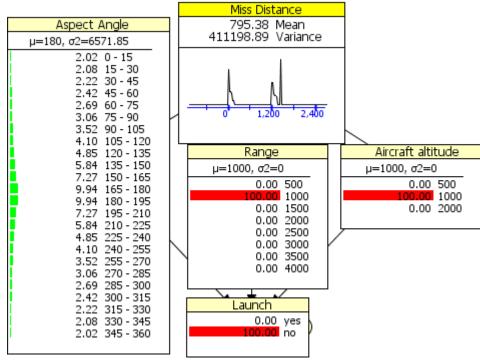
- miss distance and
- launch probability
- likely aircraft altitudes for a launch area defined by
- aspect angle and
- range





#### Bayesian network provides aggregate inference views on data





#### This example shows

- miss distance and
- likely aspect angles

for 'no launch' at given range and altitude



#### Conclusion

- Aircraft signatures can be modelled accurately
- Optical signatures are critically affected by
  - Sun reflection
  - Background radiance
- The aircraft-missile engagement is a complex series of events, many of which are only partially understood
- Bayesian Network model developed as a proof-of-concept
- Provides high-level, aggregated platform vulnerability information
- Decision Support Modelling Tools not published



#### **Further Work**

- Extend the BN model to include other variables (increase the complexity)
- These inputs to be simulated (miss distance prediction)
- Model validation
  - System specifications
  - Operational experts (broader)
- "Proper design process"
- Desktop tool



## **Thank You**

Kevin Gopaul kgopaul@csir.co.za

